

SUPPLEMENTAL TYPE CERTIFICATE

10054139

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

HELIFAB Inc.

**1318 SMEDE HWY
LAFAYETTE LA 70518
USA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.R.113

Type Certificate Holder: SIKORSKY AIRCRAFT CORPORATION

Type: S-76

Model: S-76A, S-76B, S-76C

Description of Design Change:

Installation of a 62 Gallon Auxiliary Fuel Tank in the baggage compartment in a Sikorsky S-76A, B & C Rotorcraft.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 21 July 2015



Massimo MAZZOLETTI
Head of Rotorcraft Department

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Associated Technical Documentation:

- Rotorcraft Flight Manual Supplement, document N° HF-S76-FMS-2800-0004, Revision 0, dated February 06, 2014 or later revisions of the above listed documents approved by EASA in accordance with EASA ED Decision 2004/04/CF (or subsequent revisions of this decision) and/ or the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

- Master Drawing List, document N° HF-S76-MDL-2800-0004, Revision IR, dated September 17, 2013
- Instructions for Continued Airworthiness (ICA) for Auxiliary Fuel Tank Installation, document N° HF-ICA-2800-0004, Revision IR, dated September 17, 2013.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -